CWC '19

Abdulaziz Alawad Faisal Alrashidi Naser Alrashidi Tanner Lehr Riley Sinek



Project Description

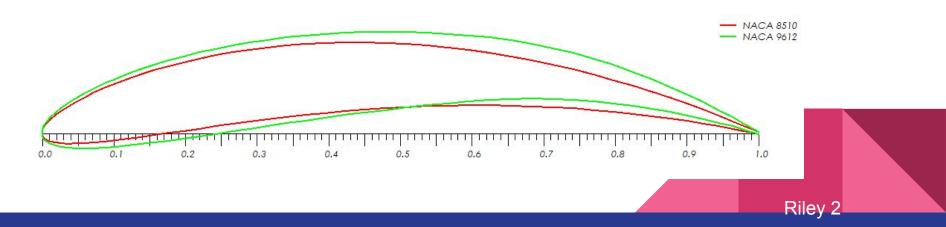
- U.S. Department of Energy is the sponsor
- **Collegiate Wind Competition**
 - Competition held in Boulder, Co. May 13th-14th Ο
 - Fifth team representing NAU at the Competition Ο
 - Collaboration with Electrical Engineers Ο





Blade Design Background

- Use wind to produce lift
 - Leads to torque around shaft, spinning the generator
- Low Reynold's Number operating environment
 - Must use a high camber airfoil to optimize lift
- Multiple airfoils implemented throughout the blade



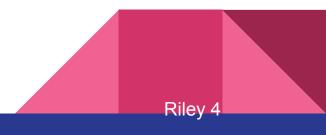
Blade Design Calculations

- Design to specified variables
 - Tip-Speed Ratio = 5
 - Blade Number = 3
- Ignore losses caused by blade and flow relationship
 - No tip loss
 - No Drag at operating attack angle
- Used MATLAB to calculate blade shape characteristics
 - Relative Wind Angle
 - Chord length
 - Twist angle



Blade Design Results

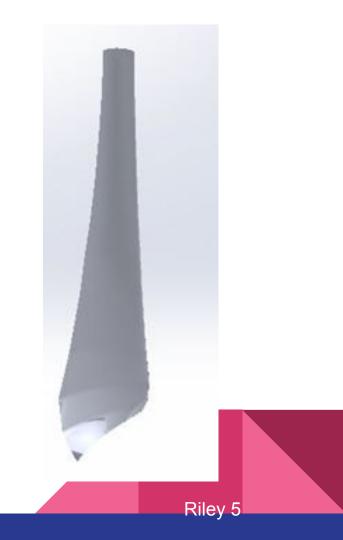
- Operating Reynold's numbers:
 - Start-up: 3,443
 - Operating: 63,396
- Q-Blade
 - Output of 30W predicted at rated wind speed
 - 25W expected
 - Start-up speed unpredictable in Q-Blade



Blade CAD

- NACA 8510 on in-board
- NACA 9612 on out-board

• Future work will investigate the performance of Selig Series airfoils for future iterations



Shaft Design Background

- Design consideration
 - Weight and size
 - Type of material

- Safety and protection



Shaft design Calculations

• Material Used for testing

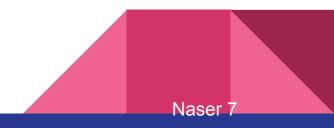
Carbon steel, alloy steel, Stainless steel

$$d = \left(\frac{32 \times n_s}{\pi \times S_y} \times \sqrt{M^2 + \frac{3}{4} T^2}\right)^{1/3}$$

- Design torque = 3.8877N.m
- Bending moment = 0.4905N.m
- Three lengths were used

L=10 cm , L=15 cm and L=20 cm

$$d = \left(\frac{32 \times n_s}{\pi \times s_y} \times \sqrt{M^2 + T^2}\right)^{1/3}$$



Shaft design Results

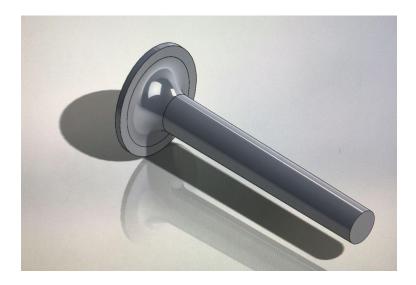
•Every material is analyzed using three different lengths and corresponding diameter are calculated using distortion energy theorem and maximum shear stress theory.

Length of shaft(cm)	Diameter of shaft(mm)	
10	12.45352	
15	12.49376	
20	12.54903	



Shaft CAD

• Make Cad model using SolidWorks.





Tower Design Assumption

- The wind velocity is constant through the tower
- Selected material that isotopic and incompressible
- Material should be linear, homogenous and elastic



Tower background and analysis

• Von mises stress for the plane

$$\sigma' = \sqrt[7]{\left(\sigma_x^2 - \sigma_x \sigma_y + \sigma_y^2 + 3\tau_{xy}^2\right)}$$

• Normal strain

$$\epsilon = \frac{l - l_o}{l_o}$$



Abdulaziz 11

Tower background and procedure

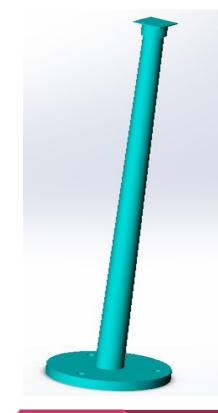
- Create base plate with 15 cm
- Bolt M10 *1.5 size
- Thickness of the plate 1.61 cm
- The length of the pipe defined as 61.91 cm include base plate and top plate

Abdulaziz 12

- Stainless steel defined in SolidWorks
 - AISI-316

Tower Cad Model

- Mesh Generation design
- Boundary conditions
- Von Mises stress result 5.314e+005 N/m²
- Deplacement analysis result 2.794e-003 mm
- Strain analysis result 2.331e-006
- FOS analysis
- \circ 3.2e+002 and maximum was achieved by 1.00e+016.





Yaw Design Background

• Yaw system is a crucial aspect in a wind turbine

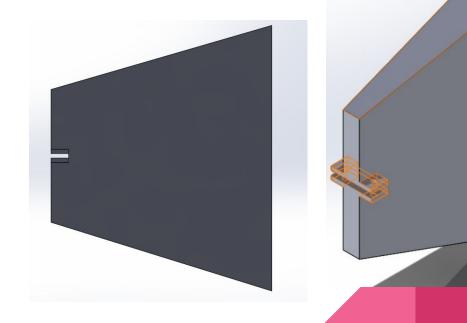
• The rotor is yawed it is less efficient compared to the non-yawed rotor

• The overall efficiency of the wind turbines is mainly influenced by a variety of factors such as wind shear, wind turbulence and yaw



Yaw Governing Equations

- r=X+2c/3 = 3.3 mm
- $M = k_N J \Omega \omega$



Brake Design Background / Calculations

- Provide sufficient force to rotor to stop the shaft from rotating
- Calculations done for Linear Actuator and Stepper Motor
- Clamping Force
 - **D** diameter, **P** operating pressure/force

$$CF=\frac{\pi D^2 P}{4}$$

- Brake Torque
 - \circ r_{e}^{-} equivalent radius, μ_{d}^{-} friction coefficient

$$BT = \frac{r_e}{2} \times (2\mu_d CF)$$



Brake Design Calculations / Results

Assumptions:

- Brake rotor material is steel
- Brake pad material is rigid molded asbestos

Friction Coef.	Rotor Diameter (mm)	Linear Actuator Force (N)	Stepper Motor Force (N)
0.36	70	18	5.5

Tanner 17

Results:

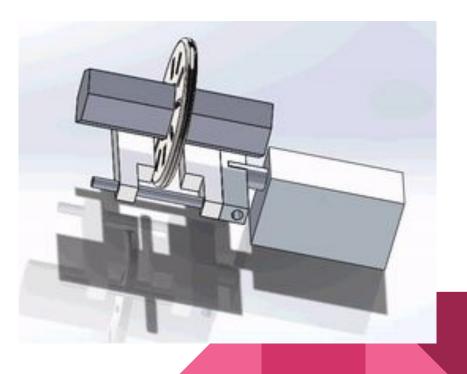
	Clamping Force (N)	Braking Torque (Nm)		
Linear Actuator	153.94	3.33		
Stepper Motor	47.04	1.02		

Brake Design CAD

• Linear actuator can apply more force

Future:

• Testing will be done to compare linear actuator and stepper motors



Tanner 18

Design / Customer Requirements

- Blades
 - Cut-in speed must be between 2.5 and 5 m/s
- Shaft
 - Withstand constant spinning from rotors
- Tower
 - \circ Less than 15 cm in diameter
- Yaw
 - Must be able to yaw 180°/sec
- Brakes
 - Must be able to stop at cut-out speed and for random tests



Budget

Table_: Project budget sheet

		Budget		
		Part	Cost	
	Bought:	Blade 2B4:C630s Blade Swashplate	\$	10.88
• Available budget		4x8x3mm Rubber Shielded Ball Bearings	\$	10.88
5		EL-Kit-003 UNO Project Super Starter Kit	\$	38.12
anticipated:		Carbonx Fiber Reinforced Nylon	\$	68.00
 \$500.00 	Future Costs:	2' of 1" OD 4130 Chromoly Steel	\$	18.29
Actual Expenses To		2' of 1" Aluminum Square Tubing	Ş	4.67
Date:		8" x 8" (0.5" thick) 6061-T6 Aluminum Plate	\$	24.93
Å 107 00		.125" 4130 Steel Sheet (12"x12")	\$	31.67
○ \$127.88		(12"x24") 6061-T6 Aluminum Sheet	\$	26.40
		SunnySky X4108S-17 KV380 Motor	\$	33.98
 Current budget does 		Z9504B 3/4" Bearing	\$	7.77
not include		PLA Filament	\$	42.00
electronics cost		Linear Actuator	\$	65.00
		1/4" 6061-T6 aluminum round	Ş	3.64
		Pillow Block Bearing	\$	16.50
		Anemometer	\$	44.95
		Assortment of nuts and bolts	\$	35.00
		Travel and Competition costs	N/A	
	Travel and Costs	s: Total:	\$	482.68

Schedule

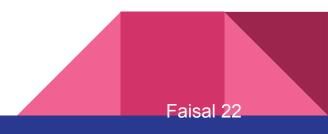
PLAN START	PLAN DURATION	ACTUAL START	ACTUAL	PERCENT COMPLETE	PERIO 1 2
11	3	24	3	100%	
15	70	24		100%	
15	4	16	2	100%	
10	5	15	2	100%	
10	10	16	10	100%	
20	6	25	10	100%	
42	30	42		100%	
26	20			15%	
24	12		1	10%	
28	10			10%	
35	8			10%	
28	8			5%	
30	5			10%	
40	5			0%	
45	5			0%	
45	12			100%	
32	40			100%	
40	5			100%	
35	10			100%	
37	5			100%	
50	10			10%	
5	40	8		100%	
45	5			0%	
	11 15 10 20 42 26 24 28 35 28 30 40 45 46 32 40 35 37 50 5	PLAN START DURATION 11 3 15 70 15 4 10 5 10 10 20 6 42 30 26 20 24 12 28 10 35 8 30 5 40 5 45 5 46 12 32 40 40 5 35 10 37 5 50 10 5 40	PLAN START DURATION START 11 3 24 15 70 24 15 70 24 15 4 16 10 5 15 10 10 16 20 6 25 42 30 42 26 20 42 26 20 42 28 10 35 30 5 40 30 5 40 45 5 46 32 40 5 35 10 37 37 5 10 35 40 8	PLAN START DURATION START DURATION 11 3 24 3 15 70 24 3 15 70 24 3 15 70 24 2 10 5 15 2 10 10 16 10 20 6 25 10 42 30 42 42 26 20 42 42 28 10 35 8 28 8 30 5 40 5 5 46 32 40 40 5 35 10 37 5 50 10 8 5	PLAN START DURATION START DURATION COMPLETE 11 3 24 3 100% 15 70 24 100% 100% 15 70 24 100% 100% 15 4 16 2 100% 10 5 15 2 100% 10 10 16 10 100% 20 6 25 10 100% 20 6 25 10 100% 24 12 30 42 10% 28 10 10% 10% 10% 28 8 5% 10% 10% 30 5 10% 10% 10% 45 5 0% 100% 10% 46 12 100% 100% 100% 35 10 100% 100% 10% 37 5 100% <

• We are on time!



Future Work

- EE team collaboration
- Begin building over Winter Break
- Begin testing in mid-February
- Upcoming assignments
 - Final Proposal Rewrite and Individual Post Mortem due 1/14
 - Website Check 1 due 1/28
 - HR1 summary and Peer Eval 1 due 2/18



Questions?